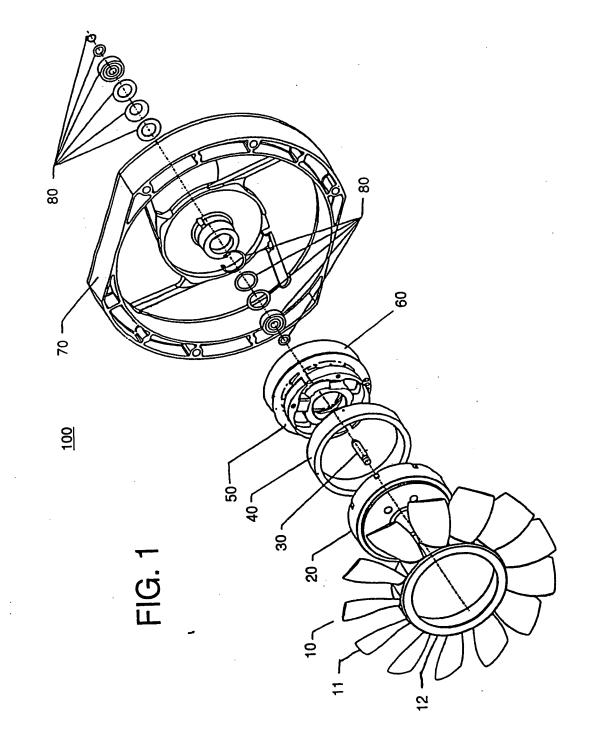
Title: AN IMPROVED IMPELLER BLADE Inventor(s): Bradbury, et al.– Express Mail Label No. EV 3200451689 US Schulte Roth & Zabel, LLP – Todd Sicklinger, Esq. Atty. Ref.: 861975/0270



<u>100</u>

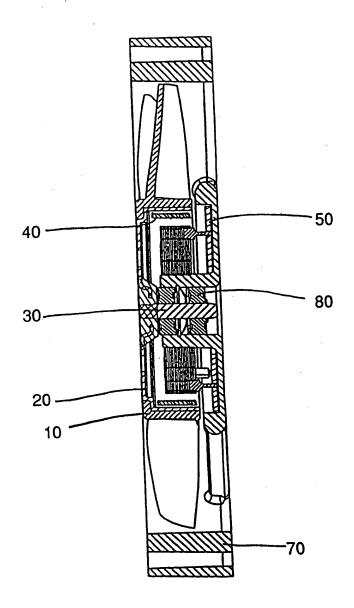


FIG. 2

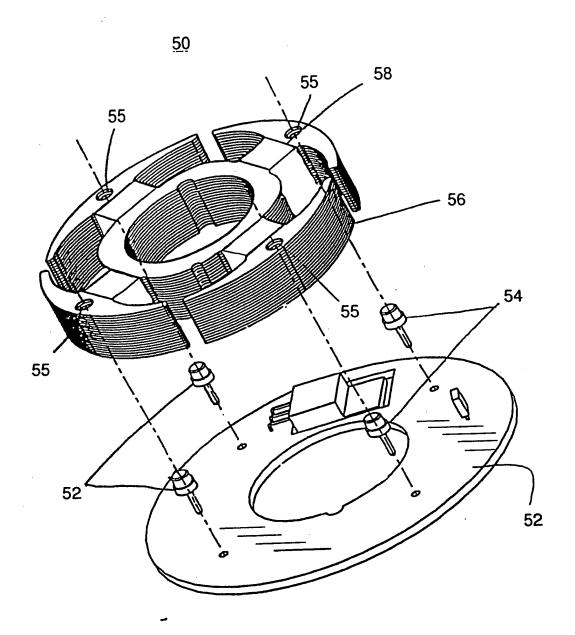
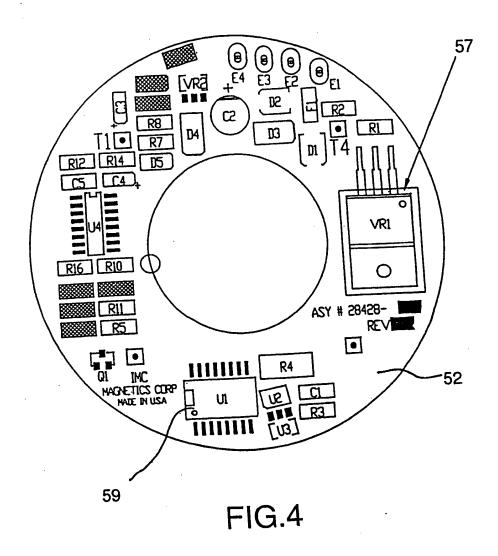
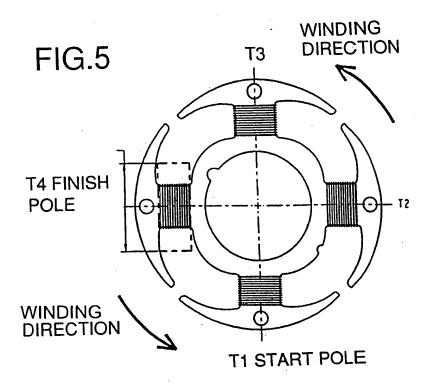
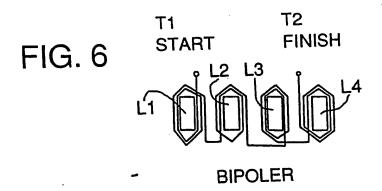


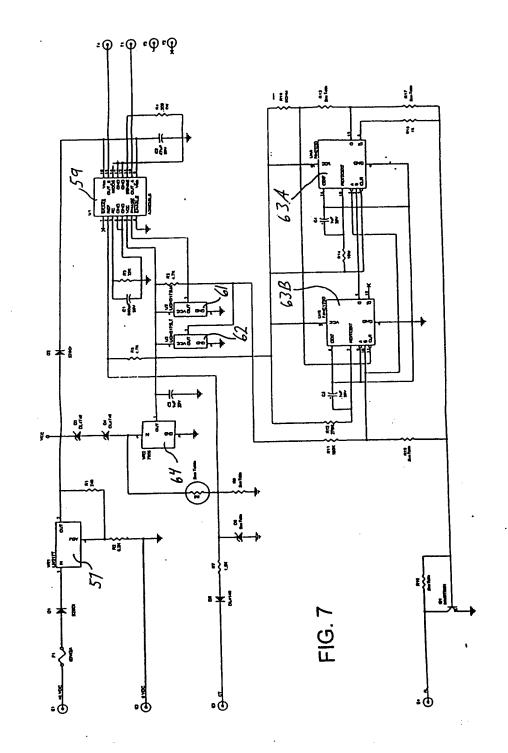
FIG. 3



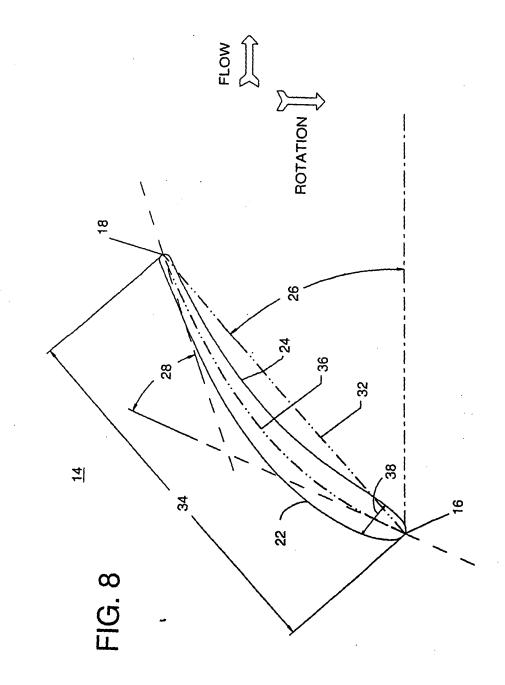




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Schulte Roth & Zabel, LLP – Todd Sicklinger, Esq.
Atty. Ref.: 861975/0270

- AXIAL STACKIN AXIS 44 FLOW T OW RADIUS AXIAL STACKING-DISTANCE (2) CIRCUMFERENTIAL STACKING DISTANCE (H) 년. root Croot ROTATION Cţi CIRCUMFERENTIAL STACKING AXIS -

FIG. 9A

FIG. 9B

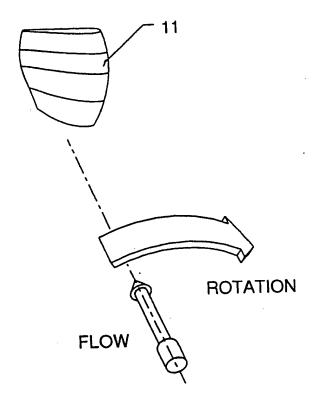


FIG. 10

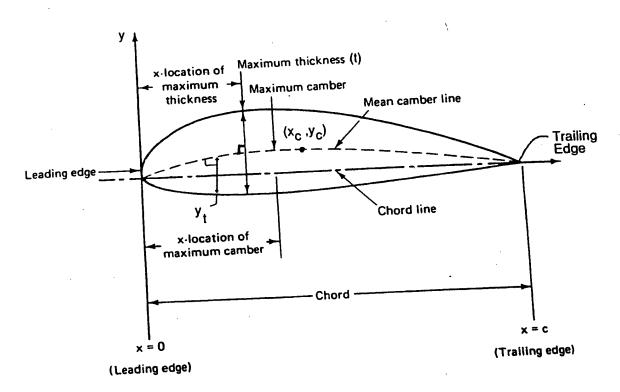


FIG. 11

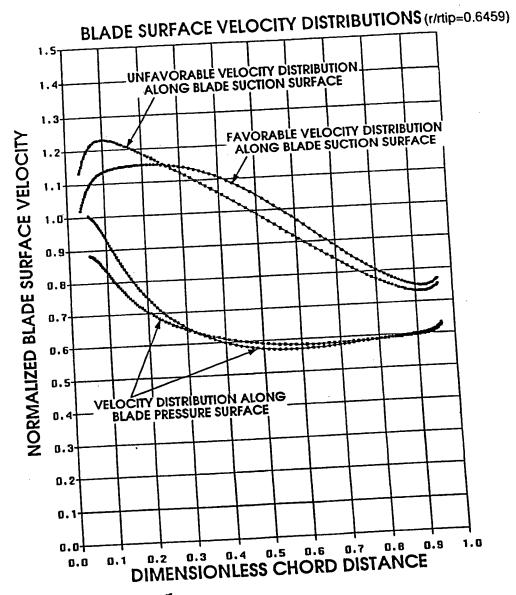


FIG. 12

#### NORMALIZED BEZIER CONTROL POINTS

		r/rtip = 0.5009			r/rtip = 0.64	150
k	Y	T	T		7 tup = 0.0	139
-	x <sub>k</sub>	y <sub>k</sub>	t <sub>k</sub>	X <sub>k</sub>	y <sub>k</sub>	t <sub>k</sub>
0	0.00	0.00000	3.66090	0.00	0.00000	2.96628
1	1.25	0.58007	5.24729	1.25	0.40469	4.25167
2	2.50	1.16014	6.46759	2.50	0.80938	5.24043
3	5.00	2.32029	8.54210	5.00	1.61876	6.92133
.4	7.50	3.48043	12.08097	7.50	2.42814	9.78873
5	10.00	4.64057	12.08097	10.00	3.23751	9.78873
6	15.00	6.96086	12.08097	15.00	4.85627	9.78873
7	20.00	9.28114	12.08097	20.00	6.47503	9.78873
8	30.00	12.34440	12.08097	30.00	9.05043	9.78873
9	40.00	12.34440	10.98270	40.00	9.05043	8.89885
10	50.00	13.37310	7.80992	50.00	9.80463	6.32807
11	60.00	12.34440	7.80992	60.00	9.05043	6.32807
12	70.00	8.22960	5.73541	70.00	6.03362	4.64718
13	80.00	6.86226	3.66090	80.00	4.64103	2.96628
14	90.00	3.48391	3.29481	90.00	2.35621	2.66965
15	95.00	1.79474	3.23379	95.00	1.21381	2.62022
16	97.50	0.95015	3.14227	97.50	0.64260	2.54606
17	98.75	0.52786	3.09651	98.75	0.35700	2.50898
18	100.00	0.00000	3.05075	100.00	0.00000	2.47190

FIG.13A

#### NORMALIZED BEZIER CONTROL POINTS

									0.00			
	T	r,	rtip = 0.79	09			r/	rtip	= 0.89			
		X <sub>k</sub>	Уk		t <sub>k</sub>	,	C <sub>k</sub>		y <sub>k</sub>		<u>t<sub>k</sub></u>	
	一十	0.00	0.00000	2.	66901	0.	00	0.0	00000		1440	
├─	0	1.25	0.28327	3.	.82559	1	.25	0.	24026	3.8	9065	
┝	1		0.56654	┼	.71526	2	.50	0.	48052	4.	79545	
L	$\frac{2}{}$	2.50	1.13309	╂	.22770	5	5.00	0.	96103	6.	33361	
<u> </u>	3	5.00	<u> </u>	╂	3.80774	1	7.50	1	.44155	8.	95753	
L	4	7.50	1.69963	+-	3.80774	┼	0.00	1	.92207	8	.95753	
L	5	10.00	2.26618	+-		┼	5.00	1	2.88310	8	.95753	
	6	15.00	3.39926	╅	8.80774	+-	20.00		3.84414	8	.95753	1
	7	20.00	4.53235		8.80774	+-	30.00		5.39486	_	3.95753	1
	8	30.00	6.54998	_	8.80774			╅	5.39486		B.14321	7
	9	40.00	6.54998	-	8.00704	+	40.00	-+-	5.8444		5.79073	7
	10	50.00	7.09582	2	5.69389	$\downarrow$	50.00	_		一十	5.79073	_
Ì	11	60.00	6.5499	В	5.69389	$\downarrow$	60.00	2	5.3948		4.25257	┪.
ł	12	70.00	4.3666	6	4.18145	$\perp$	70.0	0	3.5965			┪
	13	80.00	3.0906	1	2.66901		80.0	0	2.5388		2.71440	_
	14			8	2.4021		90.0	0	1.2889	96	2.4429	_
	15			0.80831 2.35763		3	95.0	5.00 0.664		01	2.3977	
	<b>-</b>	0.42703 2.29090		0	97.	50 0.35153		53	2.3298			
	-	10 / 27		74	2.2575	4	98.	98.75 0.195		30	2.2959	
	17	_			2.2241		100	00.0	0.000	000	2.262	00
	11	3 100.0	0.000	<u> </u>								

FIG.13B

#### NORMALIZED BEZIER CONTROL POINTS

		r/rtip = 1.0000					
k		x <sub>k</sub>		y <sub>k</sub>		t <sub>k</sub>	
0		0.00	0	.00000	3	3.06144	
1	Γ	1.25	C	.23550	4	1.38806	
2	Γ	2.50	0	.47100	٤	5.40854	
3	T	5.00	(	).94201		7.14336	
4		7.50		1.41301	1	0.10275	
5		10.00		1.88402	]	10.10275	
6	T	15.00		2.82603		10.10275	
7		20.00		3.76803	1	10.10275	
8		30.00		4.93440		10.10275	
9		40.00		4.93440		9.18432	1
10		50.00		5.34560		6.53107	1
11		60.00		4.93440	$\perp$	6.53107	1
12		70.00		3.28960	$\perp$	4.79626	4
13	13 80.00 2.47777			3.06144	4		
14	14 90.00 1.25795		5 2.75530				
15		95.00		0.64803	3	2.70427	
16	;	97.50		0.3430	3	2.62774	-
17	7	98.75		0.1906	0	2.58947	
18	3	100.0	0	0.0000	0	2.55120	)

FIG. 13C

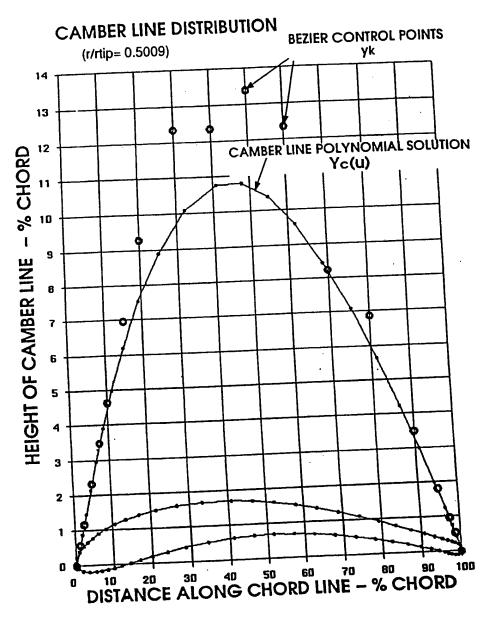
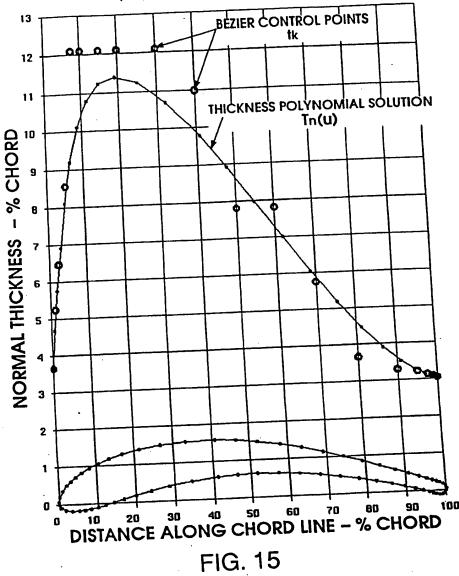
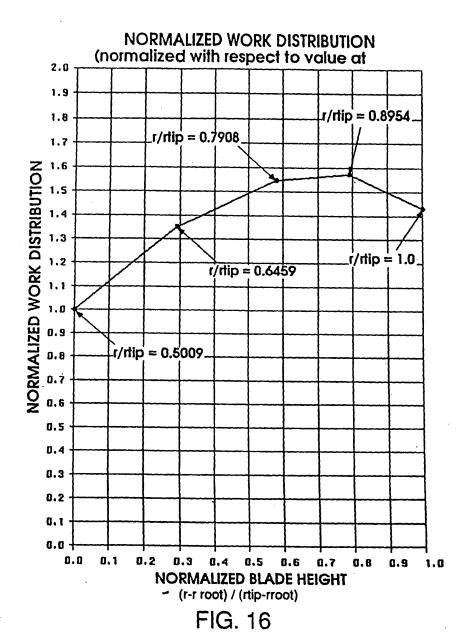


FIG. 14

### NORMAL THICKNESS DISTRIBUTION

(r/rtip=0.5009)







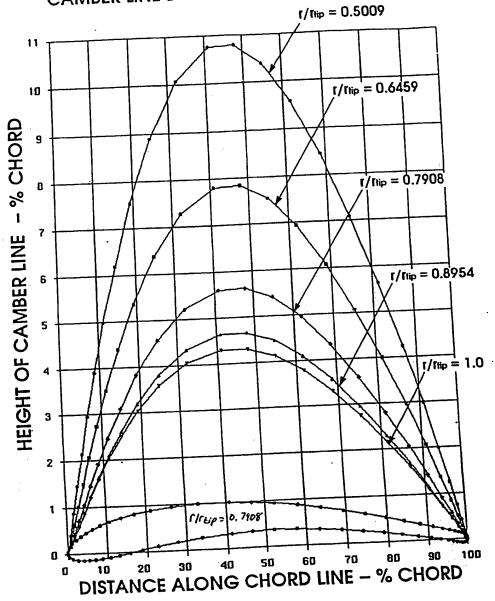


FIG. 17

### NORMAL THICKNESS DISTRIBUTION

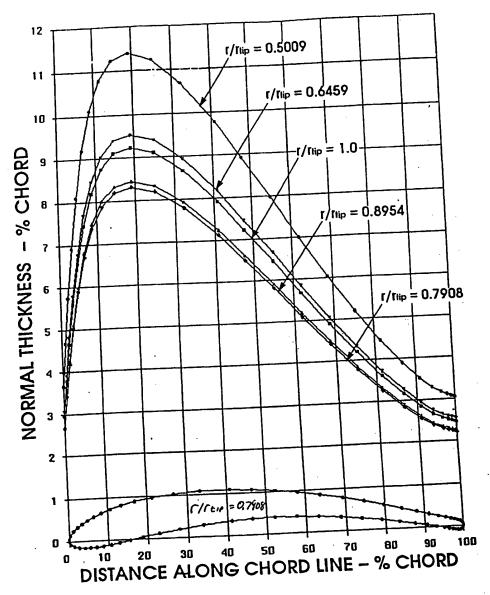


FIG. 18

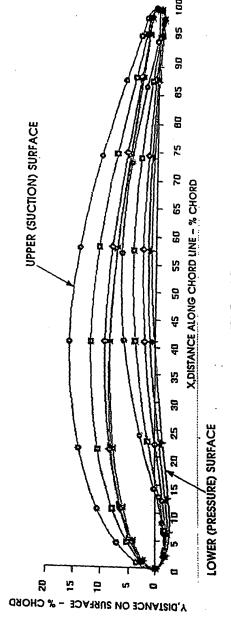
Title: AN IMPROVED IMPELLER BLADE Inventor(s): Bradbury, et al.— Express Mail Label No. EV 3200451689 US Schulte Roth & Zabel, LLP - Todd Sicklinger, Esq. Atty. Ref.: 861975/0270

NORMALALIZED BLADE PROFILES



$$a r/rtip = 0.8954$$





=1G. 1g

BLADE DEFINITION	ROOT	T			
RADIUS (INCHES)		<del> </del>		ı ————	TIP
	1.4175	1.8278	2.2381	2.5340	2.8300
NORMALIZED RADIUS	0.5009	0.6459	0.7908	0.8954	1.0000
CHORD LENGTH (INCHES)	0.7785	0.9608	1.0678	1.0499	0.9309
NORMALIZED CHORD	0.8363	1.0321	1.1471	1.1278	1.0000
ASPECT RATIO	1.8144	1.4701	1.3228	1.3454	1.5173
SOUDITY	1.1363	1.0876	0.9871	0.8573	0.6806
STAGGER ANGLE (DEGREES)	41.831	51.330	58.685	62.607	65.653
CAMBER ANGLE (DEGREES)	47.788	33.879	23.537	19.760	19.339
MAXIMUM CAMBER HEIGHT (INCHES)	0.084	0.076	0.060	0.049	0.040
MAXIMUM CAMBER HEIGHT (%CHORD)	10.823	7.863	5.652	4.671	4.320
LOCATION OF MAXIMUM CAMBER (%CHORD)	45.284	45.284	45.284	45.284	45.284
MAXIMUM THICKNESS (INCHES)	0.089	0.089	0.089	0.089	0.089
MAXIMUM THICKNESS (%CHORD)	11.392	9.230	8.305	8.446	9.526
LOCATION OF MAXIMUM THICKNESS (%CHORD)	19.174	19.174	19.174	19.174	19.174
LEADING-EDGE THICKNESS (%CHORD)	3.661	2.966	2.669	2.714	3.062
TRAILING-EDGE THICKNESS (%CHORD)	3.051	2.472	2.224	2.262	2.551
CIRCUMFERENTIAL STACKING DISTANCE (INCHES)	0.0000	0.1335	0.2141	0.2235	0.1806
NORMALIZED CIRCUMFERENTIAL STACKING DISTANÇE	0.0000	0.7392	1.1855	1.2375	1.0000
AXIAL STACKING DISTANCE (INCHES)	0.0000	-0.0419	-0.0156	0.0216	0.0800
NORMALIZED AXIAL STACKING DISTANCE	0.0000	-0.5238	-0.1950	0.2700	1.0000

FIG. 20

## NORMALIZED BLADE SURFACE COORDINATES 1/tip = 0.5009

#	XUPPER/C	YUPPER/C	XLOWER/C	YLOWER/C
1	0.00000	0.00000	0.00000	0.00000
2	-0.00164	0.00733	0.00538	-0.00793
3	0.00112	0.02029	0.01684	-0.01404
4	0.00932	0.03584	0.03360	-0.01692
5	0.02109	0.05019	0.05180	-0.01640
6	0.03931	0.06661	0.07545	-0.01370
7	0.06421	0.08411	0.10315	-0.00822
8	0.09831	0.10227	0.13646	0.00050
9	0.14433	0.12045	0.17809	0.01242
10	0.20666	0.13769	0.23280	0.02722
11	0.29321	0.15170	0.30878	0.04402
12	0.40785	0.15637	0.41136	0.05873
13	0.48494	0.15187	0.48213	0.06348
14	0.56303	0.14173	0.55559	0.06390
15	0.64006	0.12684	0.62986	0.05993
16	0.71399	0.10868	0.70267	0.05211
17	0.78256	0.08915	0.77127	0.04156
18	0.84321	0.07028	0.83245	0.02976
19	0.89333	0.05387	0.88311	0.01831
20	0.93118	0.04111	0.92126	0.00858
21	0.95695	0.03225	0.94706	0.00135
22	0.97313	0.02657	0.96313	-0.00349
23	0.98367	0.02280	0.97349	-0.00677
24	0.99251	0.01884	0.98231	-0.00908
25	0.99912	0.01295	0.99065	-0.00883
26	1.00154	0.00617	0.99676	-0.00546
27	1.00000	0.00000	1.00000	0.00000

### NORMALIZED BLADE SURFACE COORDINATES r/rtip = 0.6459

#	XUPPER/C	YUPPER/C	XLOWER/C	YLOWER/C
1	0.00000	0.00000	0.00000	0.00000
2	-0.00057	0.00593	0.00353	-0.00679
3	0.00305	0.01558	0.01204	-0.01235
4	0.01137	0.02663	0.02502	-0.01565
5	0.02239	0.03650	0.03950	-0.01647
6	0.04094	0.04910	0.06167	-0.01599
7	0.06557	0.06248	0.08862	-0.01328
8	0.09882	0.07623	0.12200	-0.00765
9	0.14379	0.08999	0.16473	0.00103
10	0.20537	0.10319	0.22185	0.01256
11	0.29210	0.11407	0.30204	0.02621
12	0.40847	0.11762	0.41074	0.03848
13	0.48605	0.11394	0.48434	0.04248
14	0.56498	0.10580	0.56036	0.04300
15	0.64310	0.09394	0.63677	0.04002
16	0.71816	0.07960	0.71120	0.03406
17 18	0.78765 0.84872	0.06441	0.78079	0.02610
19	0.89859	0.05007	0.84229	0.01738
20	0.93558	0.03800 0.02899	0.89258	0.00916
21	0.96021	0.02699	0.92982 0.95452	0.00240 -0.00245
22	0.97534	0.02299	0.96961	-0.00245
23	0.98497	0.01920	0.97912	-0.00360
24	0.99235	0.01445	0.98650	-0.00886
25	0.99813	0.01031	0.99327	-0.00797
26	1.00065	0.00511	0.99791	-0.00470
<b>27</b>	1.00000	0.00000	1.00000	0.00000
				0.0000

## NORMALIZED BLADE SURFACE COORDINATES r/rtip = 0.7908

		v 10	XLOWER/C	YLOWER/C
#	XUPPER/C	YUPPER/C	0.00000	0.00000
1	0.00000	0.00000		-0.00633
2	-0.00002	0.00532	0.00261	-0.01180
<u>3</u> .	0.00405	0.01350	0.00976	-0.01554
4	0.01246	0.02246	0.02106	-0.01720
5	0.02316	0.03013	0.03388	-0.01720 -0.01835
6	0.04207	0.04036	0.05523	
7	0.06677	0.05104	0.08164	-0.01775
8	0.09972	0.06171	0.11492	-0.01464
9	0.14420	0.07212	0.15811	-0.00877
10	0.20533	0.08191	0.21639	-0.00027
11	0.29193	0.08973	0.29865	0.01036
12	0.40884	0.09169	0.41037	0.02047
13	0.48655	0.08831	0.48539	0.02408
14	0.56581	0.08151	0.56268	0.02509
15	0.64440	0.07187	0.64012	0.02345
16	0.71997	0.06045	0.71531	0.01955
17	0.78988	0.04857	0.78535	0.01415
18	0.85113	0.03762	0.84695	0.00821
19	0.90087	0.02868	0.89702	0.00265
20	0.93743	0.02222	0.93380	-0.00185
21	0.96150	0.01808	0.95797	-0.00502
22	0.97614	0.01559	0.97261	-0.00705
23	0.98535	0.01402	0.98176	-0.00836
24	0.99210	0.01236	0.98851	-0.00894
25 25	0.99754	0.00909	0.99456	-0.00766
26	1.00018	0.00463	0.99850	-0.00438
27	1.00000	0.00000	1.00000	0.00000
Z.I	1,0000			

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# NORMALIZED BLADE SURFACE COORDINATES r/rtip = 0.8954

		V	XLOWER/C	YLOWER/C
#	XUPPER/C	YUPPER/C 0.00000	0.00000	0.00000
1	0.00000	0.00543	0.00245	-0.00654
2	0.00015		0.00954	-0.01240
3	0.00456	0.01368	0.02092	-0.01665
4	0.01340	0.02259	0.03391	-0.01884
5	0.02452	0.03007	0.05514	-0.02078
6	0.04367	0.03972	0.08145	-0.02110
7	0.06861	0.04958 0.05913	0.11469	-0.01911
8 .	0.10173	0.05913	0.15790	-0.01461
9	0.14620	0.00009	0.21624	-0.00770
10	0.20706	0.08204	0.29855	0.00125
11	0.29306	0.08262	0.41020	0.01019
12	0.40901	0.00202	0.48527	0.01374
13	0.48630	0.07269	0.56254	0.01525
14	0.56519	0.06399	0.63990	0.01463
15	0.64350 0.71889	0.05388	0.71497	0.01213
16	0.78871	0.04349	0.78489	0.00831
17	0.78671	0.03399	0.84643	0.00391
18	0.89976	0.02628	0.89652	-0.00033
19	0.93644	0.02076	0.93340	-0.00384
20 21	0.96066	0.01724	0.95769	-0.00637
22	0.97541	0.01513	0.97245	-0.00799
23	0.98470	0.01380	0.98169	-0.00906 -0.00945
24	0.99160	0.01232	0.98859	-0.00745
25 25	0.99722	0.00916	0.99472	-0.00790 -0.00450
26	1.00003	0.00471	0.99862	0.00000
27	1.00000	0.00000	1.00000	0.0000

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## NORMALIZED BLADE SURFACE COORDINATES r/rtip = 1.0000

n	V			
#	XUPPER/C	YUPPER/C	XLOWER/C	YLOWER/C
1	0.00000	0.00000	0.00000	0.00000
2	0.00018	0.00621	0.00274	-0.00746
3	0.00514	0.01586	0.01079	-0.01447
4	0.01507	0.02637	0.02371	-0.01981
5	0.02757	0.03517	0.03844	-0.02274
6	0.04710	0.04531	0.06005	-0.02523
7	0.07255	0.05535	0.08658	-0.02599
8	0.10619	0.06474	0.11987	-0.02399
9	0.15092	0.07313	0.16285	-0.02446
10	0.21140	0.08015	0.22046	-0.02036
11	0.29594	0.08465	0.30118	-0.01444
12	0.40909	0.08385	0.41012	
13	0.48526	0.07979	0.48413	0.00215
14	0.56302	0.07320	0.56031	0.00598
15	0.64027	0.06461	0.63663	0.00815
16	0.71478	0.05481	0.03003	0.00852
17	0.78401	0.03481		0.00722
18	0.84506	0.03563	0.78008 0.84138	0.00461
19	0.89512	0.03303		0.00125
20	0.93243	0.02263	0.89170	-0.00226
21	0.95741	0.01203	0.92917	-0.00536
22	0.97283	0.01507	0.95420	-0.00771
23	0.98268		0.96961	-0.00928
24	0.99047	0.01552	0.97940	-0.01035
25	0.99682	0.01387	0.98718	-0.01073
26 26	1.00001	0.01033	0.99408	-0.00901
20 27		0.00531	0.99847	-0.00508
<b>4</b> /	1.00000	0.00000	1.00000	0.00000